

DIRECTED ENERGY PROPULSION FOR INTERSTELLAR PRECURSOR MISSIONS

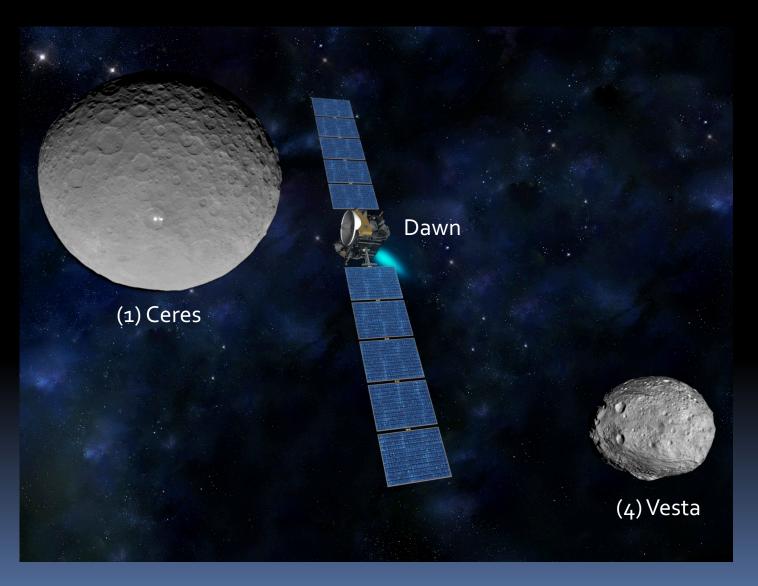
October 24, 2018

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Dawn was great but...



- The ion propulsion system on Dawn provided a record △V to the spacecraft of 11.5 km/s (25,900 mph, using just 60 gallons of fuel)
- But Dawn "only" went to ~3 AU
 - What if you wanted to go to Pluto at ~40 AU?
 - Or the solar gravity lens location at > 550 AU?
- You would have to go faster, a lot faster
 - ➤ We want △V's that are 10x higher than Dawn's, i.e., 100 to 200 km/s





- Going fast takes a lot of energy: $E = \frac{1}{2} mv^2$
- Going 10x faster takes 100x the energy

100 km/s takes ~1000x the energy density of TNT

Jet Propulsion Laboratory California Institute of Technology

- Going fast takes a lot of energy: $E = \frac{1}{2} mv^2$
- Going 10x faster takes 100x the energy
- How do you get that energy?
 - 1. You can carry it with you

Artist's concepts

Chemical propulsion

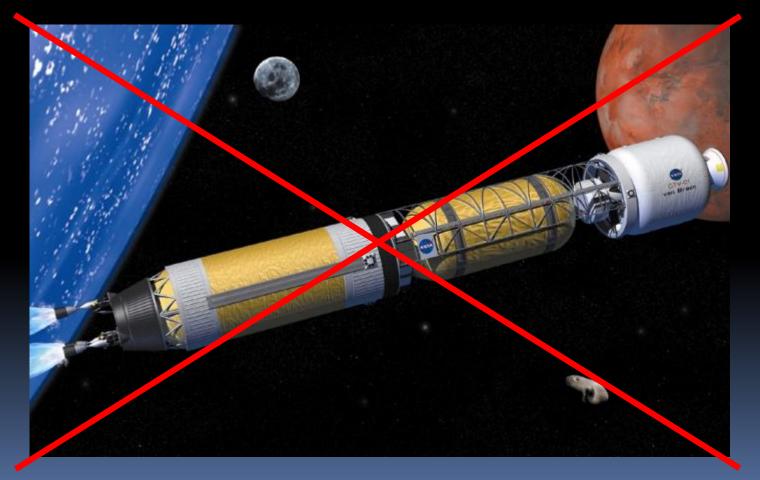
Requires too much propellant





- Going fast takes a lot of energy: $E = \frac{1}{2} mv^2$
- Going 10x faster takes 100x the energy
- How do you get that energy?
 - 1. You can carry it with you
 - Chemical propulsion
 - Nuclear propulsion
 - Nuclear thermal

Still requires too much propellant





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 - 1. You can carry it with you
 - Chemical propulsion
 - Nuclear propulsion
 - Nuclear thermal
 - Nuclear electric

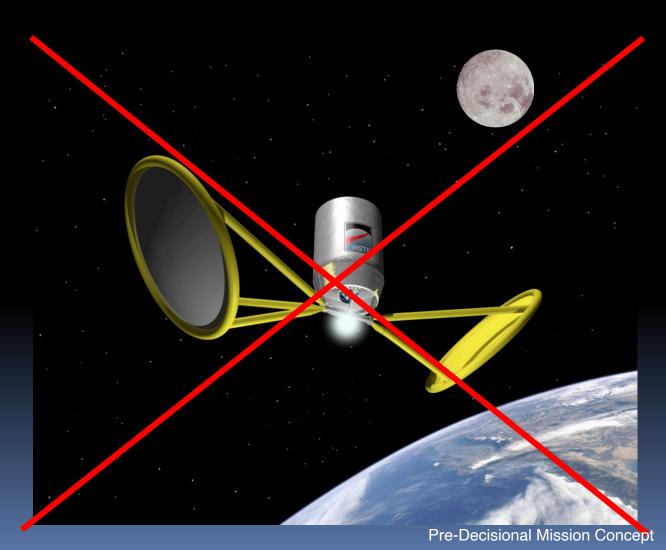
Too heavy to go fast





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- Going 10x faster takes 100x the energy
- How do you get that energy?
 - 1. You can carry it with you
 - Chemical propulsion
 - Nuclear propulsion
 - Nuclear thermal
 - Nuclear electric
 - 2. You can collect energy transmitted to you
 - Solar
 - Solar thermal

Requires too much propellant Limited solar range





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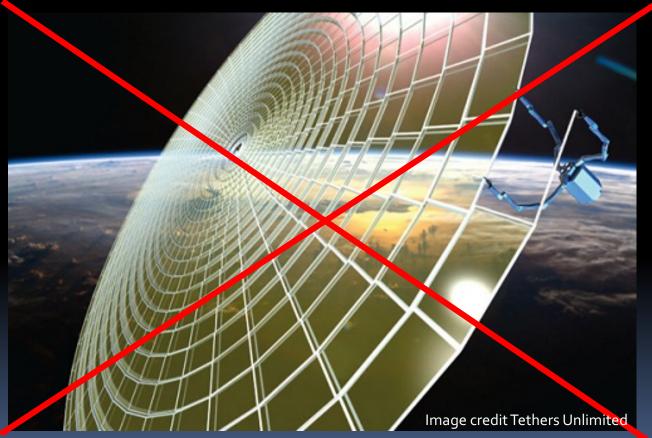
Still too much propellant Limited solar range





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 - Nuclear propulsion
 - Nuclear thermal
 - Nuclear electric
 - 2. You can collect energy transmitted to you
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 - Solar thermal
 - Solar electric
 - Microwave Sails

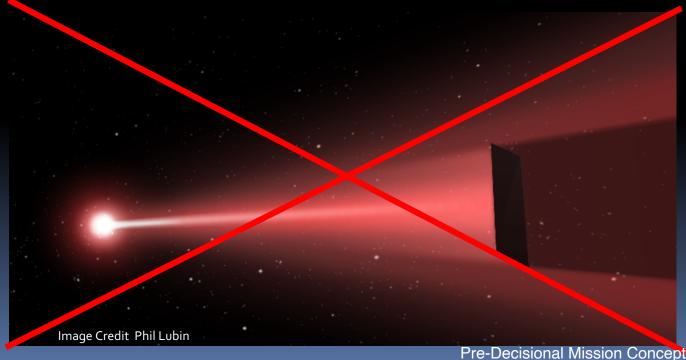
Wavelength too large >
required aperture too large





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 - Chemical propulsion
 - **Nuclear propulsion**
 - Nuclear thermal
 - Nuclear electric
 - You can collect energy transmitted to you
 - Solar
 - Solar thermal
 - Solar electric
 - Microwave
 - Laser
 - l acar cail

Thrust-to-power too low

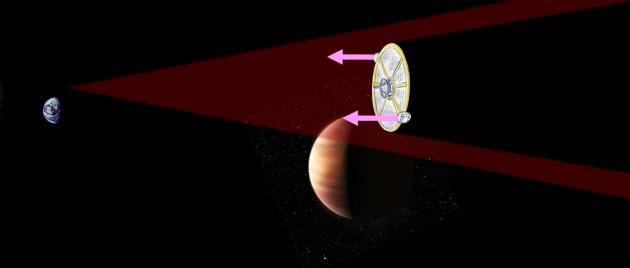




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 - Nuclear electric
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 - Solar thermal

 - Microwave
 - Laser
 - Laser sail
 - Laser electric

Yes! Laser Electric Propulsion with Ultra-high Specific Ion Engines



Three Key Features of Our Proposed Architecture to Go Fast





Kilometer-scale Laser

Don't carry the power source—laser beam power to the spacecraft

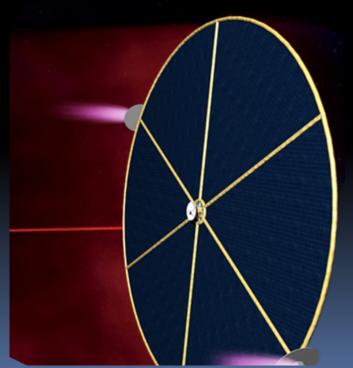






Light-weight PV Collector

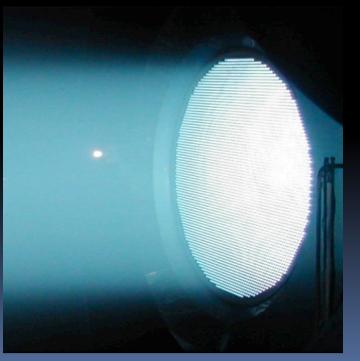
Collect the laser power and convert it to electricity to power the ion drive system





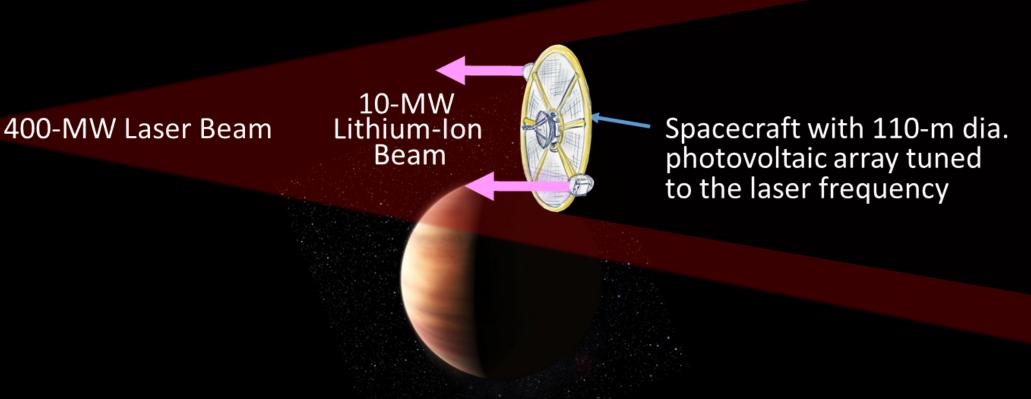
Ultra-high Isp Thruster

Increase the exhaust velocity by a factor of 10 over the best ion engines today



Artist's concepts

Pre-Decisional Mission Concept



A space-based laser beams power to a lithium-fueled, ultra-high specific impulse vehicle to enable rapid transportation throughout the solar system

Artist's concept Pre-Decisional Mission Concept

Lithium-fueled ion engines

Laser Bean

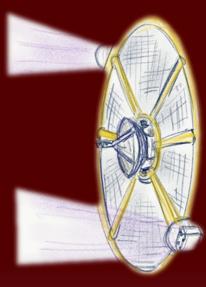
110-m diameter photovoltaic array with an areal density < 200 g/m²

Array cells tuned to the laser frequency for efficiency > 50%

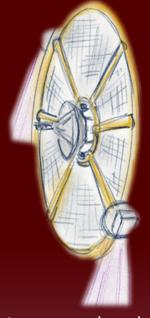
Array output voltage of 6 kV

Lithium-fueled ion engines

Laser Beam



Thrusting along the laser beam

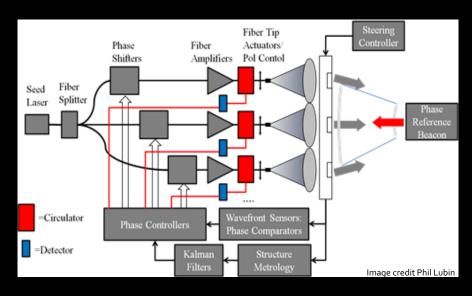


Thrusting normal to the laser beam

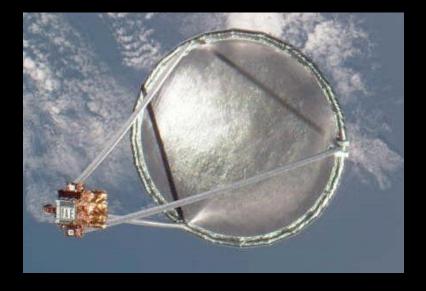
Four Technology Wavefronts



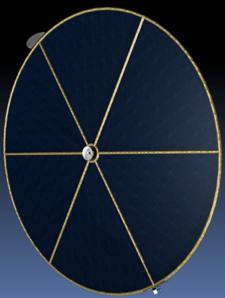
- Large Phased-Array Laser Technology
 - Space-based
 - km-scale
 - > 100's MW



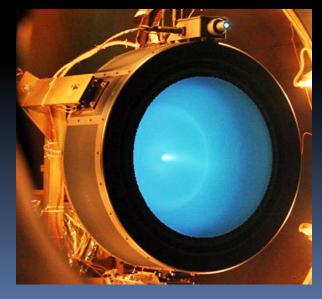
- 2. Large
 Deployable
 Structures
 - > 100-m
 - > 200 g/m²



- 3. Advanced Photovoltaics
 - > > 50% at the laser frequency
 - > Thin-film
 - ▶ 6-kV output voltage



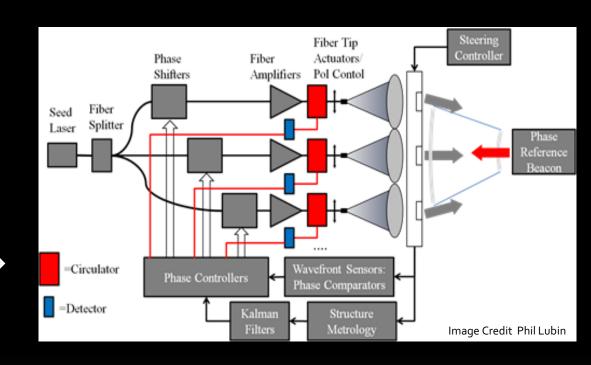
- 4. Ultra-High Specific Impulsion Ion Thrusters
 - > 50,000 s
 - > 95% efficiency
 - Direct-drive



1 Laser Scaling



- Diffraction-limited optics → very large aperture, km-scale laser
 - Phase-array laser, only known way to get a kmscale laser due to optics cost and practicality
 - Diffraction-limited performance requires a densely packed array
 - km-scale, densely packed, phased array laser → very high powers, ~100 MW
- Popular Mechanics
 - Humanity's Biggest Machines Will Be Built in Space, By Avery Thompson, Feb 16, 2018
 - "A mile-wide satellite might sound impossible, but that's exactly where the space industry is headed."



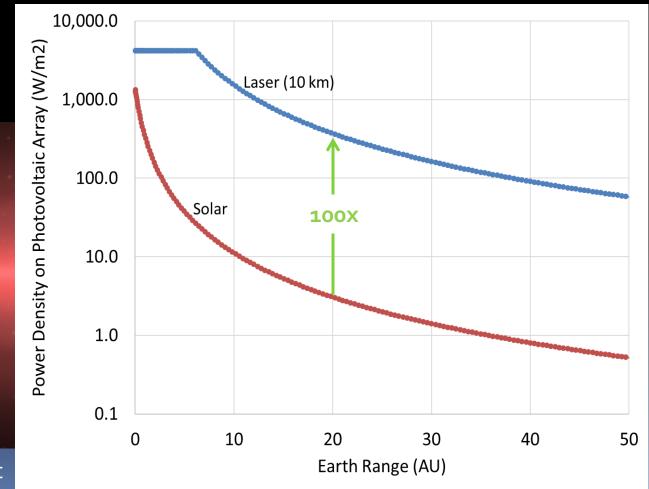
Increase Power Density by 100X Across the Solar System



High-power, space-based laser

- Phased array
- Kilometer-scale aperture
- 100's of megawatts

Beam Power Across the Solar System



mage Credit Phil Lubin

Artist's concept

Trade Laser Aperture Size, Power, and Wavelength



Option 1:

Laser Wave Length 1064 nm

Laser Aperture Dia 5000 m

Laser Output Power 800 MW

Option 2:

Laser Wave Length 300 nm

Laser Aperture Dia
 2000 m

Laser Output Power 400 MW

PV Array:

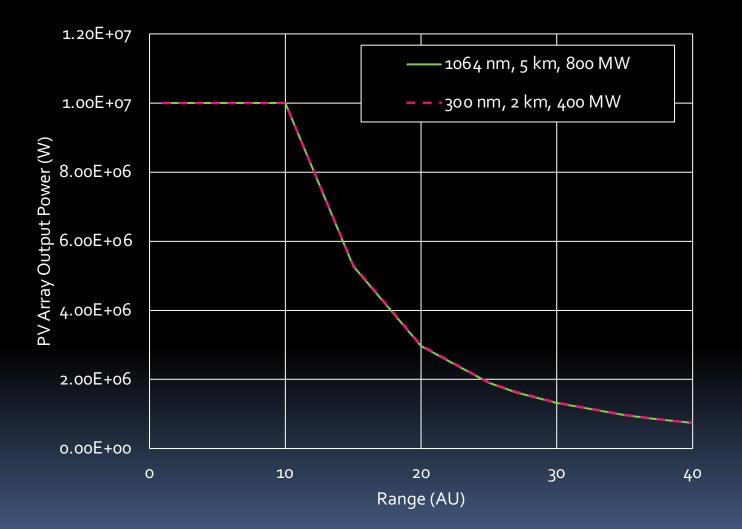
PV Array Diameter 110 m

PV Areal Density
 200 g/m²

PV Cell Efficiency 0.5

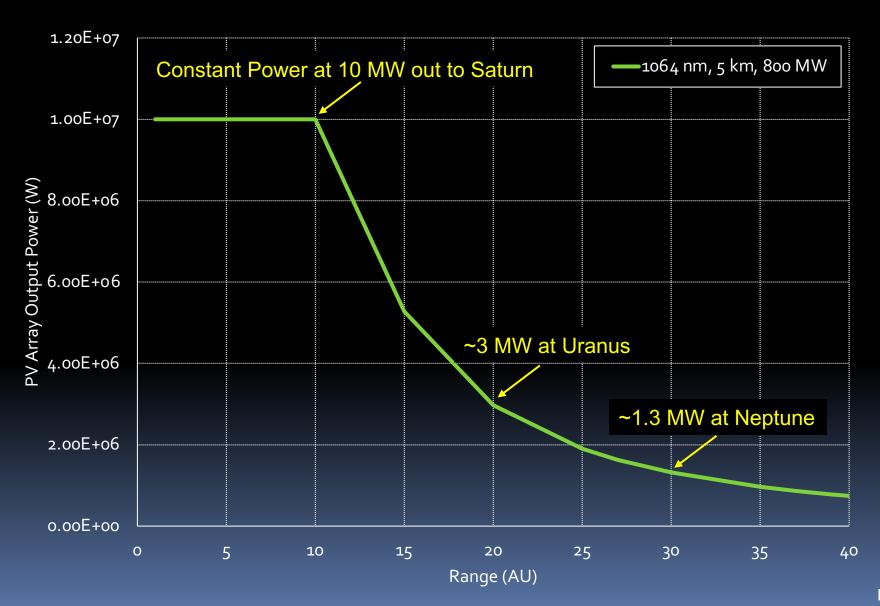
Mass of PV Array 1.90E+03 kg

Max Output Power 1.00E+07 W



System Provides Fantastic Power Levels Across the Solar System



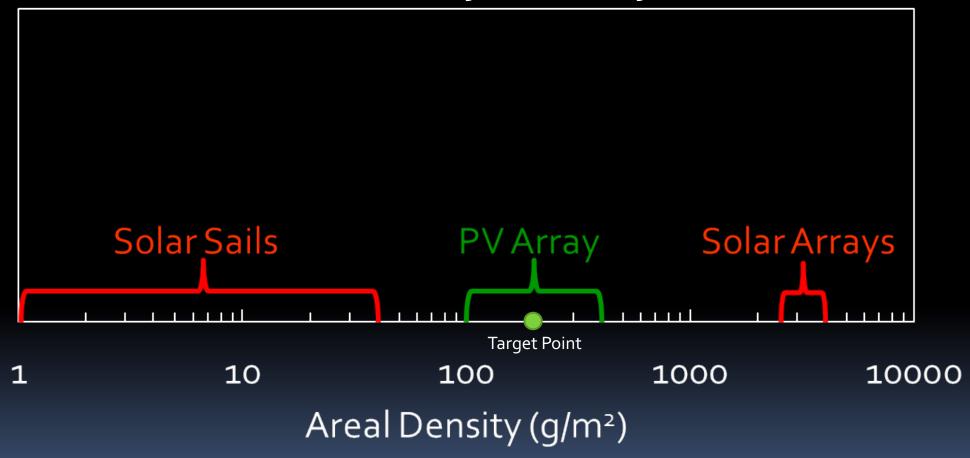




Photovoltaic Array Scaling



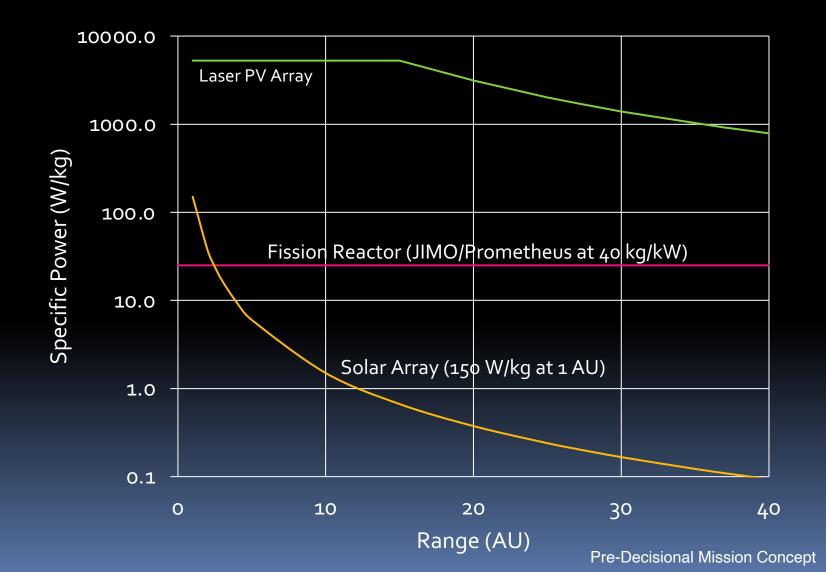
Areal Density is the Key



Provides Much Higher Specific Power (W/kg) than Other Approaches



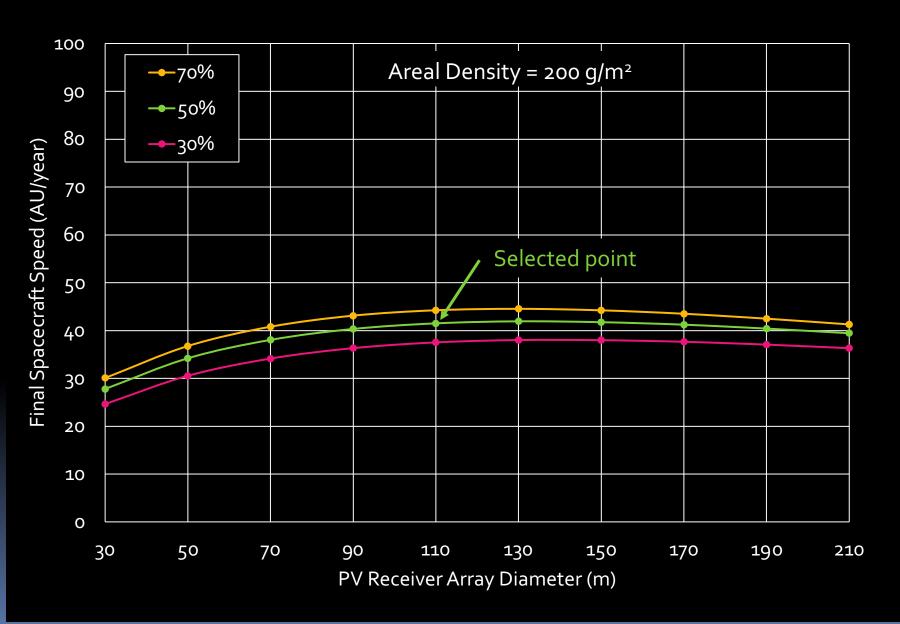
- Specific Power is the key to high performance
- Laser-driven PV array
 has significantly greater
 specific power
 throughout the solar
 system



PV Array Scaling has Broad Optimum Around 110-m dia.



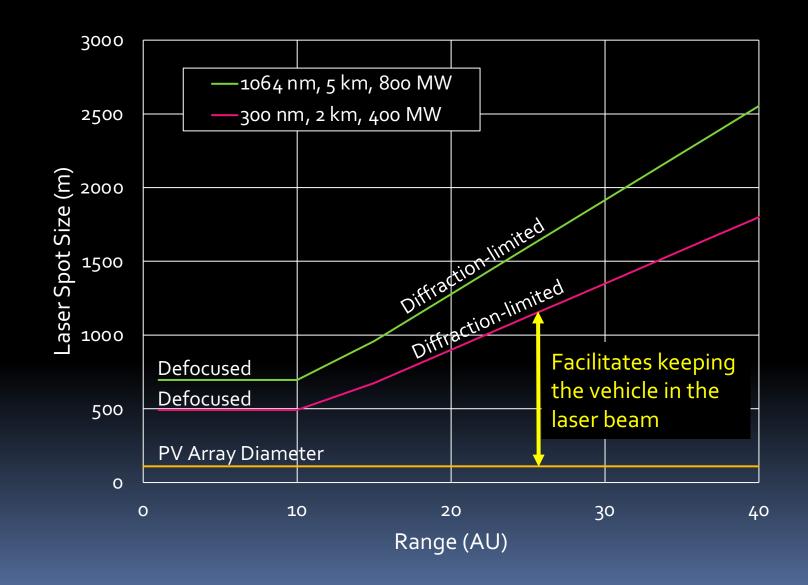
- Want 40 AU/year (190 km/s)
- Select 110-m diameter at 50%
 PV cell efficiency



Laser Spot Size is Always Bigger then the PV Array



- km-scale laser aperture required to beam power across tens' of AU
- Densely-packed phasedarray laser results in very high power, 100's of MW
- Minimize size of the EP vehicle
- Don't size the EP vehicle to use all the available power (just like outbound SEP missions!)

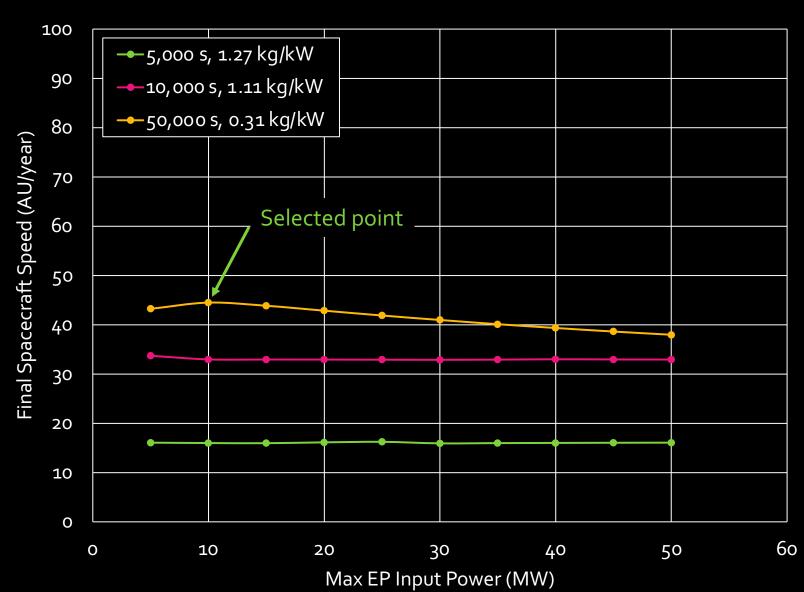




Electric Propulsion System Scaling: Isp



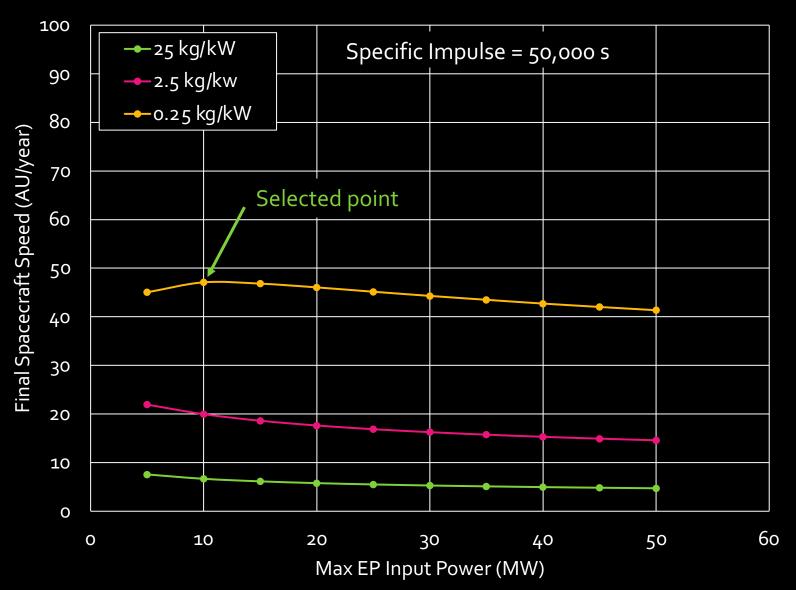
- Specific Impulses of order 50,000 s are required to achieve 40 AU/year
- Specific Masses are higher at lower Specific Impulses
- Need Max EP Power of about 10 MW



Electric Propulsion Specific Mass



Need specific masses of order 0.25 kg/kW to achieve > 40 AU/year

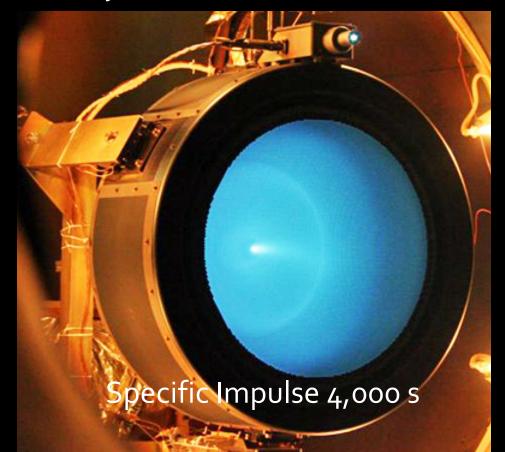


Lithium-fueled Ion Thruster



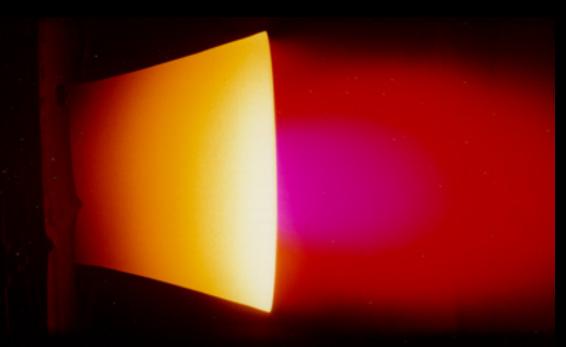
Xenon-fueled

Today's ion engines have 10X the exhaust velocity of the best chemical rockets



Lithium-fueled

Our ion engines will have 10X the exhaust velocity of the best ion thrusters

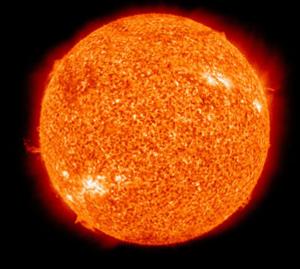


Specific Impulse > 40,000 s

What Might this Architecture Be Able to Do?



Solar Gravity Lens Mission



Human Missions to Jupiter



Pluto Orbiter Mission



Planetary Defense-Ion Beam Deflection

